



HÖGSKOLAN
I SKÖVDE

Institutionen för ingenjörsvetenskap

TENTAMEN

Kurs: Strömningsslära

Delkurs

Kurskod: FY302G

Högskolepoäng för tentamen: 2 hp

Datum: 2023-11-10

Skrivtid: 08.15 – 12.30

Ansvarig lärare: Krister Karlsson

Berörda lärare:

Hjälpmittel/bilagor

Bifogat formelblad "Formelblad – Strömningsslära FY302G". Egen räknedosa och linjal. Utdrag ur tabeller delas ut. Ring läraren vid frågor. Provformuläret ska lämnas in.

- | | |
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Tentamen omfattar sex problem och bedöms med U, G eller VG. Se dokumentet Betygskriterier.

Skrivningsresultat bör offentliggöras inom 18 arbetsdagar

Lycka till!

Antal sidor totalt

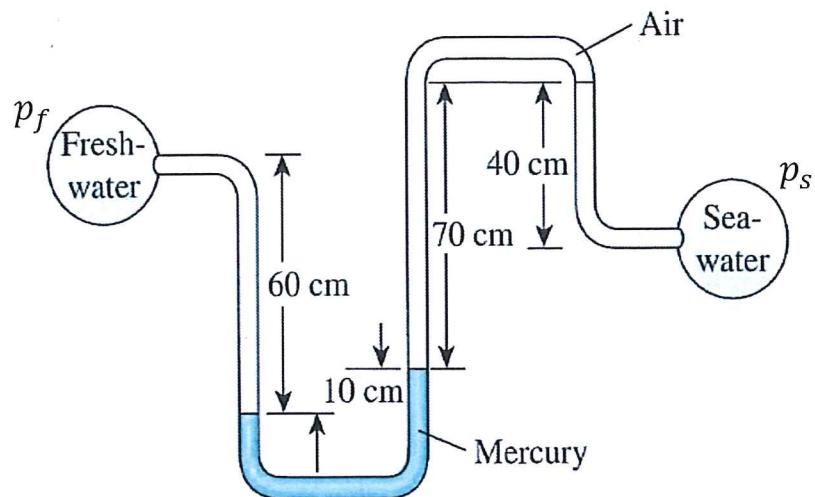
1

Basuppgift – statiskt tryck

Dricksvatten (fresh water) och saltvatten (sea water) befinner sig i två parallella ledningar och är sammankopplade med ett dubbelt U-rör enligt figuren nedan.

Bestäm tryckskillnaden mellan de två ledningarna, det vill säga $(p_f - p_s)$.

Densiteten för havsvatten i rådande fall är 1035 kg/m^3 och densiteten för kvicksilver är en faktor 13.6 större än densiteten för dricksvatten (1000 kg/m^3).

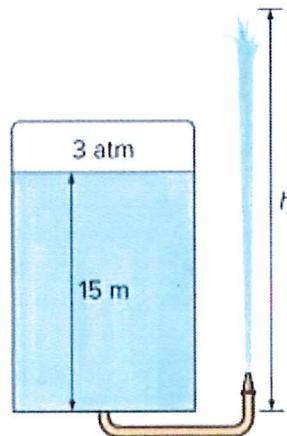


2

Basuppgift – Bernoullis ekvation

Studera en sluten tank med övertrycket 3 atm och fyllt med 15 meter vatten (se figuren).

Bestäm den maximala höjden h som vattnet kan stiga.



3

Basuppgift – rörströmning

3

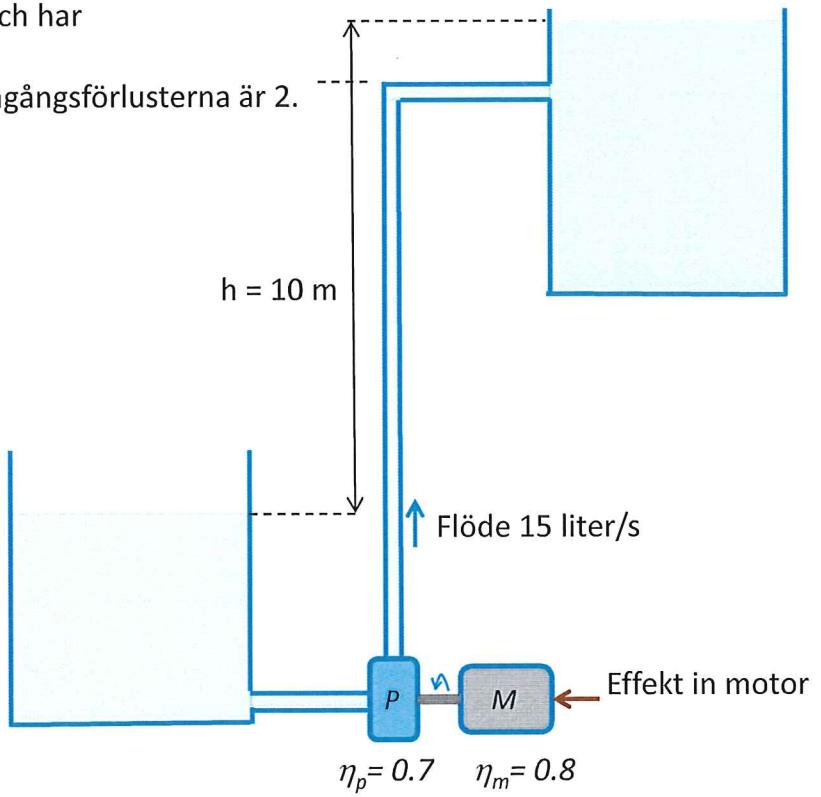
Röret mellan tankarna har innerdiametern 5.0 cm och vattenflödet är 15 liter/s.

Motorns verkningsgrad är 0.8 och pumpens 0.7. Nivån mellan vattennivåerna hålls konstant och är 10 m.

Rörmaterialet är av kommersiellt stål och har totallängden 15 m.

Vattnets temperatur är 10 °C. Totala engångsförlusterna är 2.

Beräkna effekten in till motorn.

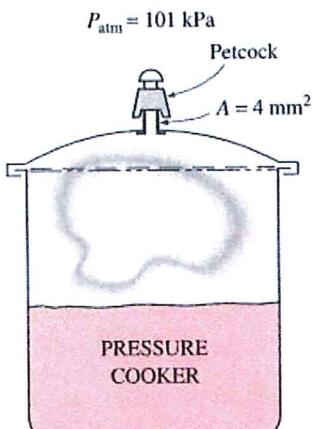


4

En tryckkokare (se figuren nedan) fungerar så att om det önskade trycket i kokaren blir för stort så lyftes ventilen (petcock) upp och släpper ut gas tills trycket åter är det önskade.

Antag att det önskade övertrycket i kokaren skall vara 100 kPa och att det normala lufttrycket är 101 kPa . Ventilens har en tvärsnittsarea på 4 mm^2 .

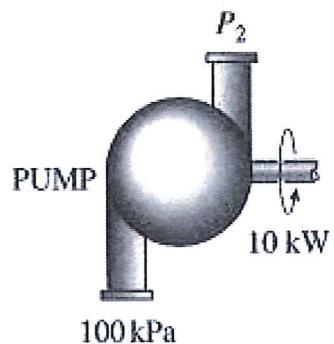
- Frilägg ventilen och rita ut alla krafterna (ventilen i vila).
- Bestäm ventilens massa (i gram).



5

Pumpen i figuren ger ett vattenflöde på 5 kilogram per sekund. Inloppet eller sugsidan (trycket 100 kPa) har likadan area som utloppet eller trycksidan (trycket p_2). Antag att vattnets densitet är 1 kilogram per liter och att pumpen ger 10 kW.

Bestäm det maximala värdet på p_2 .



6

I ett givet rörsystem finns flera ventiler, rörkrökar och kranar som ger upphov till tryckförluster (engångsförluster). Systemets sammanlagda rörlängd i är 40 meter och materialet är kommersiellt stål med diametern 5 cm. I systemet strömmar vatten med en temperatur på 15°C . En tryckdifferens på 400 kPa uppmättes mellan systemets in- och utlopp då vattenflödet var 10 liter per sekund. Utloppet ligger 2 meter högre än inloppet.

Bestäm summan av systemets engångsförluster K_L .

TABLE A-15

Properties of saturated water

Temp. <i>T</i> , °C	Saturation Pressure <i>P_{sat}</i> , kPa	Density <i>ρ</i> , kg/m ³		Enthalpy of Vaporization <i>h_{fg}</i> , kJ/kg		Specific Heat <i>c_p</i> , J/kg·K		Thermal Conductivity <i>k</i> , W/m·K		Dynamic Viscosity <i>μ</i> , kg/m·s		Prandtl Number Pr		Volume Expansion Coefficient <i>β</i> , 1/K
		Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid
0.01	0.6113	999.8	0.0048	2501	4217	1854	0.561	0.0171	1.792×10^{-3}	0.922×10^{-5}	13.5	1.00	-0.068 × 10 ⁻³	
5	0.8721	999.9	0.0068	2490	4205	1857	0.571	0.0173	1.519×10^{-3}	0.934×10^{-5}	11.2	1.00	0.015 × 10 ⁻³	
10	1.2276	999.7	0.0094	2478	4194	1862	0.580	0.0176	1.307×10^{-3}	0.946×10^{-5}	9.45	1.00	0.0733 × 10 ⁻³	
15	1.7051	999.1	0.0128	2466	4185	1863	0.589	0.0179	1.138×10^{-3}	0.959×10^{-5}	8.09	1.00	0.138 × 10 ⁻³	
20	2.339	998.0	0.0173	2454	4182	1867	0.598	0.0182	1.002×10^{-3}	0.973×10^{-5}	7.01	1.00	0.195 × 10 ⁻³	
25	3.169	997.0	0.0231	2442	4180	1870	0.607	0.0186	0.891×10^{-3}	0.987×10^{-5}	6.14	1.00	0.247 × 10 ⁻³	
30	4.246	996.0	0.0304	2431	4178	1875	0.615	0.0189	0.798×10^{-3}	1.001×10^{-5}	5.42	1.00	0.294 × 10 ⁻³	
35	5.628	994.0	0.0397	2419	4178	1880	0.623	0.0192	0.720×10^{-3}	1.016×10^{-5}	4.83	1.00	0.337 × 10 ⁻³	
40	7.384	992.1	0.0512	2407	4179	1885	0.631	0.0196	0.653×10^{-3}	1.031×10^{-5}	4.32	1.00	0.377 × 10 ⁻³	
45	9.593	990.1	0.0655	2395	4180	1892	0.637	0.0200	0.596×10^{-3}	1.046×10^{-5}	3.91	1.00	0.415 × 10 ⁻³	
50	12.35	988.1	0.0831	2383	4181	1900	0.644	0.0204	0.547×10^{-3}	1.062×10^{-5}	3.55	1.00	0.451 × 10 ⁻³	
55	15.76	985.2	0.1045	2371	4183	1908	0.649	0.0208	0.504×10^{-3}	1.077×10^{-5}	3.25	1.00	0.484 × 10 ⁻³	
60	19.94	983.3	0.1304	2359	4185	1916	0.654	0.0212	0.467×10^{-3}	1.093×10^{-5}	2.99	1.00	0.517 × 10 ⁻³	
65	25.03	980.4	0.1614	2346	4187	1926	0.659	0.0216	0.433×10^{-3}	1.110×10^{-5}	2.75	1.00	0.548 × 10 ⁻³	
70	31.19	977.5	0.1983	2334	4190	1936	0.663	0.0221	0.404×10^{-3}	1.126×10^{-5}	2.55	1.00	0.578 × 10 ⁻³	
75	38.58	974.7	0.2421	2321	4193	1948	0.667	0.0225	0.378×10^{-3}	1.142×10^{-5}	2.38	1.00	0.607 × 10 ⁻³	
80	47.39	971.8	0.2935	2309	4197	1962	0.670	0.0230	0.355×10^{-3}	1.159×10^{-5}	2.22	1.00	0.653 × 10 ⁻³	
85	57.83	968.1	0.3536	2296	4201	1977	0.673	0.0235	0.333×10^{-3}	1.176×10^{-5}	2.08	1.00	0.670 × 10 ⁻³	
90	70.14	965.3	0.4235	2283	4206	1993	0.675	0.0240	0.315×10^{-3}	1.193×10^{-5}	1.96	1.00	0.702 × 10 ⁻³	
95	84.55	961.5	0.5045	2270	4212	2010	0.677	0.0246	0.297×10^{-3}	1.210×10^{-5}	1.85	1.00	0.716 × 10 ⁻³	
100	101.33	957.9	0.5978	2257	4217	2029	0.679	0.0251	0.282×10^{-3}	1.227×10^{-5}	1.75	1.00	0.750 × 10 ⁻³	
110	143.27	950.6	0.8263	2230	4229	2071	0.682	0.0262	0.255×10^{-3}	1.261×10^{-5}	1.58	1.00	0.798 × 10 ⁻³	
120	198.53	943.4	1.121	2203	4244	2120	0.683	0.0275	0.232×10^{-3}	1.296×10^{-5}	1.44	1.00	0.858 × 10 ⁻³	
130	270.1	934.6	1.496	2174	4263	2177	0.684	0.0288	0.213×10^{-3}	1.330×10^{-5}	1.33	1.01	0.913 × 10 ⁻³	
140	361.3	921.7	1.965	2145	4286	2244	0.683	0.0301	0.197×10^{-3}	1.365×10^{-5}	1.24	1.02	0.970 × 10 ⁻³	
150	475.8	916.6	2.546	2114	4311	2314	0.682	0.0316	0.183×10^{-3}	1.399×10^{-5}	1.16	1.02	1.025 × 10 ⁻³	
160	617.8	907.4	3.256	2083	4340	2420	0.680	0.0331	0.170×10^{-3}	1.434×10^{-5}	1.09	1.05	1.145 × 10 ⁻³	
170	791.7	897.7	4.119	2050	4370	2490	0.677	0.0347	0.160×10^{-3}	1.468×10^{-5}	1.03	1.05	1.178 × 10 ⁻³	
180	1,002.1	887.3	5.153	2015	4410	2590	0.673	0.0364	0.150×10^{-3}	1.502×10^{-5}	0.983	1.07	1.210 × 10 ⁻³	
190	1,254.4	876.4	6.388	1979	4460	2710	0.669	0.0382	0.142×10^{-3}	1.537×10^{-5}	0.947	1.09	1.280 × 10 ⁻³	
200	1,553.8	864.3	7.852	1941	4500	2840	0.663	0.0401	0.134×10^{-3}	1.571×10^{-5}	0.910	1.11	1.350 × 10 ⁻³	
220	2,318	840.3	11.60	1859	4610	3110	0.650	0.0442	0.122×10^{-3}	1.641×10^{-5}	0.865	1.15	1.520 × 10 ⁻³	
240	3,344	813.7	16.73	1767	4760	3520	0.632	0.0487	0.111×10^{-3}	1.712×10^{-5}	0.836	1.24	1.720 × 10 ⁻³	
260	4,688	783.7	23.69	1663	4970	4070	0.609	0.0540	0.102×10^{-3}	1.788×10^{-5}	0.832	1.35	2.000 × 10 ⁻³	
280	6,412	750.8	33.15	1544	5280	4835	0.581	0.0605	0.094×10^{-3}	1.870×10^{-5}	0.854	1.49	2.380 × 10 ⁻³	
300	8,581	713.8	46.15	1405	5750	5980	0.548	0.0695	0.086×10^{-3}	1.965×10^{-5}	0.902	1.69	2.950 × 10 ⁻³	
320	11,274	667.1	64.57	1239	6540	7900	0.509	0.0836	0.078×10^{-3}	2.084×10^{-5}	1.00	1.97		
340	14,586	610.5	92.62	1028	8240	11,870	0.469	0.110	0.070×10^{-3}	2.255×10^{-5}	1.23	2.43		
360	18,651	528.3	144.0	720	14,690	25,800	0.427	0.178	0.060×10^{-3}	2.571×10^{-5}	2.06	3.73		
374.14	22,090	317.0	317.0	0	—	—	—	—	0.043×10^{-3}	4.313×10^{-5}				

Note 1: Kinematic viscosity ν and thermal diffusivity α can be calculated from their definitions, $\nu = \mu/\rho$ and $\alpha = k/\rho c_p = \nu/\text{Pr}$. The temperatures 0.01°C, 100°C, and 374.14°C are the triple-, boiling-, and critical-point temperatures of water, respectively. The properties listed above (except the vapor density) can be used at any pressure with negligible error except at temperatures near the critical-point value.

Note 2: The unit kJ/kg·°C for specific heat is equivalent to kJ/kg·K, and the unit W/m·°C for thermal conductivity is equivalent to W/m·K.

Source: Viscosity and thermal conductivity data are from J. V. Sengers and J. T. R. Watson, *Journal of Physical and Chemical Reference Data* 15 (1986), pp. 1291–1322. Other data are obtained from various sources or calculated.

TABLE A-16

Properties of saturated refrigerant-134a

Temp., °C	Saturation Pressure, P, kPa	Density ρ, kg/m³		Enthalpy of Vaporization h _{fg} , kJ/kg		Specific Heat c _p , J/kg·K		Thermal Conductivity k, W/m·K		Dynamic Viscosity μ, kg/m·s		Prandtl Number Pr		Volume Expansion Coefficient β, 1/K		Surface Tension, N/m	
		Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	N/m
-40	51.2	1418	2.773	225.9	1254	748.6	0.1101	0.00811	4.878 × 10⁻⁴	2.550 × 10⁻⁶	5.558	0.235	0.00205	0.01760			
-35	66.2	1403	3.524	222.7	1264	764.1	0.1084	0.00862	4.509 × 10⁻⁴	3.003 × 10⁻⁶	5.257	0.266	0.00209	0.01682			
-30	84.4	1389	4.429	219.5	1273	780.2	0.1066	0.00913	4.178 × 10⁻⁴	3.504 × 10⁻⁶	4.992	0.299	0.00215	0.01604			
-25	106.5	1374	5.509	216.3	1283	797.2	0.1047	0.00963	3.882 × 10⁻⁴	4.054 × 10⁻⁶	4.757	0.335	0.00220	0.01527			
-20	132.8	1359	6.787	213.0	1294	814.9	0.1028	0.01013	3.614 × 10⁻⁴	4.651 × 10⁻⁶	4.548	0.374	0.00227	0.01451			
-15	164.0	1343	8.288	209.5	1306	833.5	0.1009	0.01063	3.371 × 10⁻⁴	5.295 × 10⁻⁶	4.363	0.415	0.00233	0.01376			
-10	200.7	1327	10.04	206.0	1318	853.1	0.0989	0.01112	3.150 × 10⁻⁴	5.982 × 10⁻⁶	4.198	0.459	0.00241	0.01302			
-5	243.5	1311	12.07	202.4	1330	873.8	0.0968	0.01161	2.947 × 10⁻⁴	6.709 × 10⁻⁶	4.051	0.505	0.00249	0.01229			
0	293.0	1295	14.42	198.7	1344	895.6	0.0947	0.01210	2.761 × 10⁻⁴	7.471 × 10⁻⁶	3.919	0.553	0.00258	0.01156			
5	349.9	1278	17.12	194.8	1358	918.7	0.0925	0.01259	2.589 × 10⁻⁴	8.264 × 10⁻⁶	3.802	0.603	0.00269	0.01084			
10	414.9	1261	20.22	190.8	1374	943.2	0.0903	0.01308	2.430 × 10⁻⁴	9.081 × 10⁻⁶	3.697	0.655	0.00280	0.01014			
15	488.7	1244	23.75	186.6	1390	969.4	0.0880	0.01357	2.281 × 10⁻⁴	9.915 × 10⁻⁶	3.604	0.708	0.00293	0.00944			
20	572.1	1226	27.77	182.3	1408	997.6	0.0856	0.01406	2.142 × 10⁻⁴	1.075 × 10⁻⁵	3.521	0.763	0.00307	0.00876			
25	665.8	1207	32.34	177.8	1427	1028	0.0833	0.01456	2.012 × 10⁻⁴	1.160 × 10⁻⁵	3.448	0.819	0.00324	0.00808			
30	770.6	1188	37.53	173.1	1448	1061	0.0808	0.01507	1.888 × 10⁻⁴	1.244 × 10⁻⁵	3.383	0.877	0.00342	0.00742			
35	887.5	1168	43.41	168.2	1471	1098	0.0783	0.01558	1.772 × 10⁻⁴	1.327 × 10⁻⁵	3.328	0.935	0.00364	0.00677			
40	1017.1	1147	50.08	163.0	1498	1138	0.0757	0.01610	1.660 × 10⁻⁴	1.408 × 10⁻⁵	3.285	0.995	0.00390	0.00613			
45	1160.5	1125	57.66	157.6	1529	1184	0.0731	0.01664	1.554 × 10⁻⁴	1.486 × 10⁻⁵	3.253	1.058	0.00420	0.00550			
50	1318.6	1102	66.27	151.8	1566	1237	0.0704	0.01720	1.453 × 10⁻⁴	1.562 × 10⁻⁵	3.231	1.123	0.00455	0.00489			
55	1492.3	1078	76.11	145.7	1608	1298	0.0676	0.01777	1.355 × 10⁻⁴	1.634 × 10⁻⁵	3.223	1.193	0.00500	0.00429			
60	1682.8	1053	87.38	139.1	1659	1372	0.0647	0.01838	1.260 × 10⁻⁴	1.704 × 10⁻⁵	3.229	1.272	0.00554	0.00372			
65	1891.0	1026	100.4	132.1	1722	1462	0.0618	0.01902	1.167 × 10⁻⁴	1.771 × 10⁻⁵	3.255	1.362	0.00624	0.00315			
70	2118.2	996.2	115.6	124.4	1801	1577	0.0587	0.01972	1.077 × 10⁻⁴	1.839 × 10⁻⁵	3.307	1.471	0.00716	0.00261			
75	2365.8	964	133.6	115.9	1907	1731	0.0555	0.02048	9.891 × 10⁻⁵	1.908 × 10⁻⁵	3.400	1.612	0.00843	0.00209			
80	2635.2	928.2	155.3	106.4	2056	1948	0.0521	0.02133	9.011 × 10⁻⁵	1.982 × 10⁻⁵	3.558	1.810	0.01031	0.00160			
85	2928.2	887.1	182.3	95.4	2287	2281	0.0484	0.02233	8.124 × 10⁻⁵	2.071 × 10⁻⁵	3.837	2.116	0.01336	0.00114			
90	3246.9	837.7	217.8	82.2	2701	2865	0.0444	0.02357	7.203 × 10⁻⁵	2.187 × 10⁻⁵	4.385	2.658	0.01911	0.00071			
95	3594.1	772.5	269.3	64.9	3675	4144	0.0396	0.02544	6.190 × 10⁻⁵	2.370 × 10⁻⁵	5.746	3.862	0.03343	0.00033			
100	3975.1	651.7	376.3	33.9	7959	8785	0.0322	0.02989	4.765 × 10⁻⁵	2.833 × 10⁻⁵	11.77	8.326	0.10047	0.00004			

Note 1: Kinematic viscosity ν and thermal diffusivity α can be calculated from their definitions, $\nu = \mu/\rho$ and $\alpha = k/\rho c_p = \nu/\text{Pr}$. The properties listed here (except the vapor density) can be used at any pressures with negligible error except at temperatures near the critical-point value.

Note 2: The unit kJ/kg·°C for specific heat is equivalent to kJ/kg·K, and the unit W/m·°C for thermal conductivity is equivalent to W/m·K.

Source: Data generated from the EES software developed by S. A. Klein and F. L. Alvarado. Original sources: R. Tillner-Roth and H. D. Baehr, "An International Standard Formulation for the Thermodynamic Properties of 1,1,1,2-Tetrafluoroethane (HFC-134a) for Temperatures from 170 K to 455 K and Pressures up to 70 MPa," *J. Phys. Chem. Ref. Data*, Vol. 23, No. 5, 1994; M.J. Assael, N. K. Dalaouti, A. A. Griva, and J. H. Dymond, "Viscosity and Thermal Conductivity of Halogenated Methane and Ethane Refrigerants," *JR*, Vol. 22, pp. 525–535, 1999; NIST REFPROP 6 program (M. O. McLinden, S. A. Klein, E. W. Lemmon, and A. P. Peskin, Physical and Chemical Properties Division, National Institute of Standards and Technology, Boulder, CO 80303, 1995).

TABLE A-17

Properties of saturated ammonia

Temp. <i>T</i> , °C	Saturation Pressure <i>P</i> , kPa	Density <i>ρ</i> , kg/m ³		Enthalpy of Vaporization <i>h_{f_p}</i> , kJ/kg		Specific Heat <i>c_{p_p}</i> , J/kg·K		Thermal Conductivity <i>k</i> , W/m·K		Dynamic Viscosity <i>μ</i> , kg/m·s		Prandtl Number Pr		Volume Expansion Coefficient <i>β</i> , 1/K		Surface Tension, N/m	
		Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor	Liquid	Vapor
-40	71.66	690.2	0.6435	1389	4414	2242	—	0.01792	2.926 × 10 ⁻⁴	7.957 × 10 ⁻⁶	—	0.9955	0.00176	0.03565			
-30	119.4	677.8	1.037	1360	4465	2322	—	0.01898	2.630 × 10 ⁻⁴	8.311 × 10 ⁻⁶	—	1.017	0.00185	0.03341			
-25	151.5	671.5	1.296	1345	4489	2369	0.5968	0.01957	2.492 × 10 ⁻⁴	8.490 × 10 ⁻⁶	1.875	1.028	0.00190	0.03229			
-20	190.1	665.1	1.603	1329	4514	2420	0.5853	0.02015	2.361 × 10 ⁻⁴	8.669 × 10 ⁻⁶	1.821	1.041	0.00194	0.03118			
-15	236.2	658.6	1.966	1313	4538	2476	0.5737	0.02075	2.236 × 10 ⁻⁴	8.851 × 10 ⁻⁶	1.769	1.056	0.00199	0.03007			
-10	290.8	652.1	2.391	1297	4564	2536	0.5621	0.02138	2.117 × 10 ⁻⁴	9.034 × 10 ⁻⁶	1.718	1.072	0.00205	0.02896			
-5	354.9	645.4	2.886	1280	4589	2601	0.5505	0.02203	2.003 × 10 ⁻⁴	9.218 × 10 ⁻⁶	1.670	1.089	0.00210	0.02786			
0	429.6	638.6	3.458	1262	4617	2672	0.5390	0.02270	1.896 × 10 ⁻⁴	9.405 × 10 ⁻⁶	1.624	1.107	0.00216	0.02676			
5	516	631.7	4.116	1244	4645	2749	0.5274	0.02341	1.794 × 10 ⁻⁴	9.593 × 10 ⁻⁶	1.580	1.126	0.00223	0.02566			
10	615.3	624.6	4.870	1226	4676	2831	0.5158	0.02415	1.697 × 10 ⁻⁴	9.784 × 10 ⁻⁶	1.539	1.147	0.00230	0.02457			
15	728.8	617.5	5.729	1206	4709	2920	0.5042	0.02492	1.606 × 10 ⁻⁴	9.978 × 10 ⁻⁶	1.500	1.169	0.00237	0.02348			
20	857.8	610.2	6.705	1186	4745	3016	0.4927	0.02573	1.519 × 10 ⁻⁴	1.017 × 10 ⁻⁵	1.463	1.193	0.00245	0.02240			
25	1003	602.8	7.809	1166	4784	3120	0.4811	0.02658	1.438 × 10 ⁻⁴	1.037 × 10 ⁻⁵	1.430	1.218	0.00254	0.02132			
30	1167	595.2	9.055	1144	4828	3232	0.4695	0.02748	1.361 × 10 ⁻⁴	1.057 × 10 ⁻⁵	1.399	1.244	0.00264	0.02024			
35	1351	587.4	10.46	1122	4877	3354	0.4579	0.02843	1.288 × 10 ⁻⁴	1.078 × 10 ⁻⁵	1.372	1.272	0.00275	0.01917			
40	1555	579.4	12.03	1099	4932	3486	0.4464	0.02943	1.219 × 10 ⁻⁴	1.099 × 10 ⁻⁵	1.347	1.303	0.00287	0.01810			
45	1782	571.3	13.8	1075	4993	3631	0.4348	0.03049	1.155 × 10 ⁻⁴	1.121 × 10 ⁻⁵	1.327	1.335	0.00301	0.01704			
50	2033	562.9	15.78	1051	5063	3790	0.4232	0.03162	1.094 × 10 ⁻⁴	1.143 × 10 ⁻⁵	1.310	1.371	0.00316	0.01598			
55	2310	554.2	18.00	1025	5143	3967	0.4116	0.03283	1.037 × 10 ⁻⁴	1.166 × 10 ⁻⁵	1.297	1.409	0.00334	0.01493			
60	2614	545.2	20.48	997.4	5234	4163	0.4001	0.03412	9.846 × 10 ⁻⁵	1.189 × 10 ⁻⁵	1.288	1.452	0.00354	0.01389			
65	2948	536.0	23.26	968.9	5340	4384	0.3885	0.03550	9.347 × 10 ⁻⁵	1.213 × 10 ⁻⁵	1.285	1.499	0.00377	0.01285			
70	3312	526.3	26.39	939.0	5463	4634	0.3769	0.03700	8.879 × 10 ⁻⁵	1.238 × 10 ⁻⁵	1.287	1.551	0.00404	0.01181			
75	3709	516.2	29.90	907.5	5608	4923	0.3653	0.03862	8.440 × 10 ⁻⁵	1.264 × 10 ⁻⁵	1.296	1.612	0.00436	0.01079			
80	4141	505.7	33.87	874.1	5780	5260	0.3538	0.04038	8.030 × 10 ⁻⁵	1.292 × 10 ⁻⁵	1.312	1.683	0.00474	0.00977			
85	4609	494.5	38.36	838.6	5988	5659	0.3422	0.04232	7.646 × 10 ⁻⁵	1.322 × 10 ⁻⁵	1.338	1.768	0.00521	0.00876			
90	5116	482.8	43.48	800.6	6242	6142	0.3306	0.04447	7.284 × 10 ⁻⁵	1.354 × 10 ⁻⁵	1.375	1.871	0.00579	0.00776			
95	5665	470.2	49.35	759.8	6561	6740	0.3190	0.04687	6.946 × 10 ⁻⁵	1.389 × 10 ⁻⁵	1.429	1.999	0.00652	0.00677			
100	6257	456.6	56.15	715.5	6972	7503	0.3075	0.04958	6.628 × 10 ⁻⁵	1.429 × 10 ⁻⁵	1.503	2.163	0.00749	0.00579			

Note 1: Kinematic viscosity ν and thermal diffusivity α can be calculated from their definitions, $\nu = \mu/\rho$ and $\alpha = k/c_p = \nu/\text{Pr}$. The properties listed here (except the vapor density) can be used at any pressures with negligible error except at temperatures near the critical-point value.

Note 2: The unit kJ/kg·°C for specific heat is equivalent to J/kg·K, and the unit W/m·°C for thermal conductivity is equivalent to W/m·K.

Source: Data generated from the EES software developed by S. A. Klein and F. L. Alvarado. Original sources: Tillner-Roth, Harms-Watzenberg, and Baehr, "Eine neue Fundamentalgleichung für Ammoniak," DKV-Tagungsbericht 20:167–181, 1993; Liley and Desal, "Thermophysical Properties of Refrigerants," ASHRAE, 1993, ISBN 1-1883413-10-9.

TABLE A-19

Properties of liquids

Temp. <i>T</i> , °C	Density <i>ρ</i> , kg/m ³	Specific Heat <i>c_p</i> , J/kg·K	Thermal Conductivity <i>k</i> , W/m·K	Thermal Diffusivity <i>α</i> , m ² /s	Dynamic Viscosity <i>μ</i> , kg/m·s	Kinematic Viscosity <i>ν</i> , m ² /s	Prandtl Number <i>Pr</i>	Volume Expansion Coeff. <i>β</i> , 1/K
<i>Methane [CH₄]</i>								
-160	420.2	3492	0.1863	1.270 × 10 ⁻⁷	1.133 × 10 ⁻⁴	2.699 × 10 ⁻⁷	2.126	0.00352
-150	405.0	3580	0.1703	1.174 × 10 ⁻⁷	9.169 × 10 ⁻⁵	2.264 × 10 ⁻⁷	1.927	0.00391
-140	388.8	3700	0.1550	1.077 × 10 ⁻⁷	7.551 × 10 ⁻⁵	1.942 × 10 ⁻⁷	1.803	0.00444
-130	371.1	3875	0.1402	9.749 × 10 ⁻⁸	6.288 × 10 ⁻⁵	1.694 × 10 ⁻⁷	1.738	0.00520
-120	351.4	4146	0.1258	8.634 × 10 ⁻⁸	5.257 × 10 ⁻⁵	1.496 × 10 ⁻⁷	1.732	0.00637
-110	328.8	4611	0.1115	7.356 × 10 ⁻⁸	4.377 × 10 ⁻⁵	1.331 × 10 ⁻⁷	1.810	0.00841
-100	301.0	5578	0.0967	5.761 × 10 ⁻⁸	3.577 × 10 ⁻⁵	1.188 × 10 ⁻⁷	2.063	0.01282
-90	261.7	8902	0.0797	3.423 × 10 ⁻⁸	2.761 × 10 ⁻⁵	1.055 × 10 ⁻⁷	3.082	0.02922
<i>Methanol [CH₃(OH)]</i>								
20	788.4	2515	0.1987	1.002 × 10 ⁻⁷	5.857 × 10 ⁻⁴	7.429 × 10 ⁻⁷	7.414	0.00118
30	779.1	2577	0.1980	9.862 × 10 ⁻⁸	5.088 × 10 ⁻⁴	6.531 × 10 ⁻⁷	6.622	0.00120
40	769.6	2644	0.1972	9.690 × 10 ⁻⁸	4.460 × 10 ⁻⁴	5.795 × 10 ⁻⁷	5.980	0.00123
50	760.1	2718	0.1965	9.509 × 10 ⁻⁸	3.942 × 10 ⁻⁴	5.185 × 10 ⁻⁷	5.453	0.00127
60	750.4	2798	0.1957	9.320 × 10 ⁻⁸	3.510 × 10 ⁻⁴	4.677 × 10 ⁻⁷	5.018	0.00132
70	740.4	2885	0.1950	9.128 × 10 ⁻⁸	3.146 × 10 ⁻⁴	4.250 × 10 ⁻⁷	4.655	0.00137
<i>Isobutane (R600a)</i>								
-100	683.8	1881	0.1383	1.075 × 10 ⁻⁷	9.305 × 10 ⁻⁴	1.360 × 10 ⁻⁶	12.65	0.00142
-75	659.3	1970	0.1357	1.044 × 10 ⁻⁷	5.624 × 10 ⁻⁴	8.531 × 10 ⁻⁷	8.167	0.00150
-50	634.3	2069	0.1283	9.773 × 10 ⁻⁸	3.769 × 10 ⁻⁴	5.942 × 10 ⁻⁷	6.079	0.00161
-25	608.2	2180	0.1181	8.906 × 10 ⁻⁸	2.688 × 10 ⁻⁴	4.420 × 10 ⁻⁷	4.963	0.00177
0	580.6	2306	0.1068	7.974 × 10 ⁻⁸	1.993 × 10 ⁻⁴	3.432 × 10 ⁻⁷	4.304	0.00199
25	550.7	2455	0.0956	7.069 × 10 ⁻⁸	1.510 × 10 ⁻⁴	2.743 × 10 ⁻⁷	3.880	0.00232
50	517.3	2640	0.0851	6.233 × 10 ⁻⁸	1.155 × 10 ⁻⁴	2.233 × 10 ⁻⁷	3.582	0.00286
75	478.5	2896	0.0757	5.460 × 10 ⁻⁸	8.785 × 10 ⁻⁵	1.836 × 10 ⁻⁷	3.363	0.00385
100	429.6	3361	0.0669	4.634 × 10 ⁻⁸	6.483 × 10 ⁻⁵	1.509 × 10 ⁻⁷	3.256	0.00628
<i>Glycerin</i>								
0	1276	2262	0.2820	9.773 × 10 ⁻⁸	10.49	8.219 × 10 ⁻³	84,101	
5	1273	2288	0.2835	9.732 × 10 ⁻⁸	6.730	5.287 × 10 ⁻³	54,327	
10	1270	2320	0.2846	9.662 × 10 ⁻⁸	4.241	3.339 × 10 ⁻³	34,561	
15	1267	2354	0.2856	9.576 × 10 ⁻⁸	2.496	1.970 × 10 ⁻³	20,570	
20	1264	2386	0.2860	9.484 × 10 ⁻⁸	1.519	1.201 × 10 ⁻³	12,671	
25	1261	2416	0.2860	9.388 × 10 ⁻⁸	0.9934	7.878 × 10 ⁻⁴	8,392	
30	1258	2447	0.2860	9.291 × 10 ⁻⁸	0.6582	5.232 × 10 ⁻⁴	5,631	
35	1255	2478	0.2860	9.195 × 10 ⁻⁸	0.4347	3.464 × 10 ⁻⁴	3,767	
40	1252	2513	0.2863	9.101 × 10 ⁻⁸	0.3073	2.455 × 10 ⁻⁴	2,697	
<i>Engine Oil (unused)</i>								
0	899.0	1797	0.1469	9.097 × 10 ⁻⁸	3.814	4.242 × 10 ⁻³	46,636	0.00070
20	888.1	1881	0.1450	8.680 × 10 ⁻⁸	0.8374	9.429 × 10 ⁻⁴	10,863	0.00070
40	876.0	1964	0.1444	8.391 × 10 ⁻⁸	0.2177	2.485 × 10 ⁻⁴	2,962	0.00070
60	863.9	2048	0.1404	7.934 × 10 ⁻⁸	0.07399	8.565 × 10 ⁻⁵	1,080	0.00070
80	852.0	2132	0.1380	7.599 × 10 ⁻⁸	0.03232	3.794 × 10 ⁻⁵	499.3	0.00070
100	840.0	2220	0.1367	7.330 × 10 ⁻⁸	0.01718	2.046 × 10 ⁻⁵	279.1	0.00070
120	828.9	2308	0.1347	7.042 × 10 ⁻⁸	0.01029	1.241 × 10 ⁻⁵	176.3	0.00070
140	816.8	2395	0.1330	6.798 × 10 ⁻⁸	0.006558	8.029 × 10 ⁻⁶	118.1	0.00070
150	810.3	2441	0.1327	6.708 × 10 ⁻⁸	0.005344	6.595 × 10 ⁻⁶	98.31	0.00070

Source: Data generated from the EES software developed by S. A. Klein and F. L. Alvarado. Originally based on various sources.

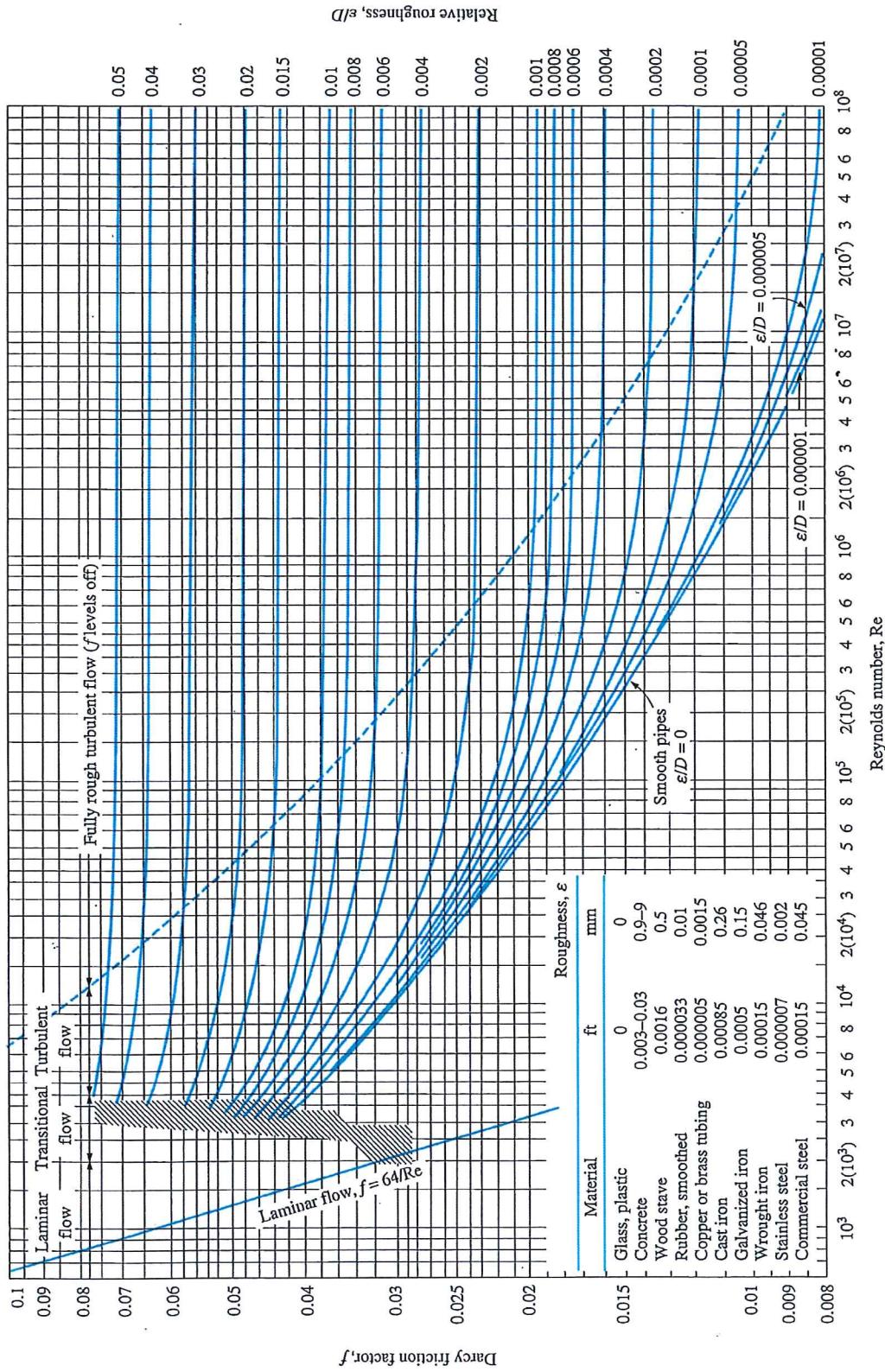
TABLE A-22

Properties of air at 1 atm pressure

Temp. <i>T</i> , °C	Density <i>p</i> , kg/m³	Specific Heat <i>c_p</i> , J/kg·K	Thermal Conductivity <i>k</i> , W/m·K	Thermal Diffusivity <i>α</i> , m²/s	Dynamic Viscosity <i>μ</i> , kg/m·s	Kinematic Viscosity <i>ν</i> , m²/s	Prandtl Number <i>Pr</i>
-150	2.866	983	0.01171	4.158×10^{-6}	8.636×10^{-6}	3.013×10^{-6}	0.7246
-100	2.038	966	0.01582	8.036×10^{-6}	1.189×10^{-5}	5.837×10^{-6}	0.7263
-50	1.582	999	0.01979	1.252×10^{-5}	1.474×10^{-5}	9.319×10^{-6}	0.7440
-40	1.514	1002	0.02057	1.356×10^{-5}	1.527×10^{-5}	1.008×10^{-5}	0.7436
-30	1.451	1004	0.02134	1.465×10^{-5}	1.579×10^{-5}	1.087×10^{-5}	0.7425
-20	1.394	1005	0.02211	1.578×10^{-5}	1.630×10^{-5}	1.169×10^{-5}	0.7408
-10	1.341	1006	0.02288	1.696×10^{-5}	1.680×10^{-5}	1.252×10^{-5}	0.7387
0	1.292	1006	0.02364	1.818×10^{-5}	1.729×10^{-5}	1.338×10^{-5}	0.7362
5	1.269	1006	0.02401	1.880×10^{-5}	1.754×10^{-5}	1.382×10^{-5}	0.7350
10	1.246	1006	0.02439	1.944×10^{-5}	1.778×10^{-5}	1.426×10^{-5}	0.7336
15	1.225	1007	0.02476	2.009×10^{-5}	1.802×10^{-5}	1.470×10^{-5}	0.7323
20	1.204	1007	0.02514	2.074×10^{-5}	1.825×10^{-5}	1.516×10^{-5}	0.7309
25	1.184	1007	0.02551	2.141×10^{-5}	1.849×10^{-5}	1.562×10^{-5}	0.7296
30	1.164	1007	0.02588	2.208×10^{-5}	1.872×10^{-5}	1.608×10^{-5}	0.7282
35	1.145	1007	0.02625	2.277×10^{-5}	1.895×10^{-5}	1.655×10^{-5}	0.7268
40	1.127	1007	0.02662	2.346×10^{-5}	1.918×10^{-5}	1.702×10^{-5}	0.7255
45	1.109	1007	0.02699	2.416×10^{-5}	1.941×10^{-5}	1.750×10^{-5}	0.7241
50	1.092	1007	0.02735	2.487×10^{-5}	1.963×10^{-5}	1.798×10^{-5}	0.7228
60	1.059	1007	0.02808	2.632×10^{-5}	2.008×10^{-5}	1.896×10^{-5}	0.7202
70	1.028	1007	0.02881	2.780×10^{-5}	2.052×10^{-5}	1.995×10^{-5}	0.7177
80	0.9994	1008	0.02953	2.931×10^{-5}	2.096×10^{-5}	2.097×10^{-5}	0.7154
90	0.9718	1008	0.03024	3.086×10^{-5}	2.139×10^{-5}	2.201×10^{-5}	0.7132
100	0.9458	1009	0.03095	3.243×10^{-5}	2.181×10^{-5}	2.306×10^{-5}	0.7111
120	0.8977	1011	0.03235	3.565×10^{-5}	2.264×10^{-5}	2.522×10^{-5}	0.7073
140	0.8542	1013	0.03374	3.898×10^{-5}	2.345×10^{-5}	2.745×10^{-5}	0.7041
160	0.8148	1016	0.03511	4.241×10^{-5}	2.420×10^{-5}	2.975×10^{-5}	0.7014
180	0.7788	1019	0.03646	4.593×10^{-5}	2.504×10^{-5}	3.212×10^{-5}	0.6992
200	0.7459	1023	0.03779	4.954×10^{-5}	2.577×10^{-5}	3.455×10^{-5}	0.6974
250	0.6746	1033	0.04104	5.890×10^{-5}	2.760×10^{-5}	4.091×10^{-5}	0.6946
300	0.6158	1044	0.04418	6.871×10^{-5}	2.934×10^{-5}	4.765×10^{-5}	0.6935
350	0.5664	1056	0.04721	7.892×10^{-5}	3.101×10^{-5}	5.475×10^{-5}	0.6937
400	0.5243	1069	0.05015	8.951×10^{-5}	3.261×10^{-5}	6.219×10^{-5}	0.6948
450	0.4880	1081	0.05298	1.004×10^{-4}	3.415×10^{-5}	6.997×10^{-5}	0.6965
500	0.4565	1093	0.05572	1.117×10^{-4}	3.563×10^{-5}	7.806×10^{-5}	0.6986
600	0.4042	1115	0.06093	1.352×10^{-4}	3.846×10^{-5}	9.515×10^{-5}	0.7037
700	0.3627	1135	0.06581	1.598×10^{-4}	4.111×10^{-5}	1.133×10^{-4}	0.7092
800	0.3289	1153	0.07037	1.855×10^{-4}	4.362×10^{-5}	1.326×10^{-4}	0.7149
900	0.3008	1169	0.07465	2.122×10^{-4}	4.600×10^{-5}	1.529×10^{-4}	0.7206
1000	0.2772	1184	0.07868	2.398×10^{-4}	4.826×10^{-5}	1.741×10^{-4}	0.7260
1500	0.1990	1234	0.09599	3.908×10^{-4}	5.817×10^{-5}	2.922×10^{-4}	0.7478
2000	0.1553	1264	0.11113	5.664×10^{-4}	6.630×10^{-5}	4.270×10^{-4}	0.7539

Note: For ideal gases, the properties c_p , k , μ , and Pr are independent of pressure. The properties p , ν , and α at a pressure P (in atm) other than 1 atm are determined by multiplying the values of p at the given temperature by P and dividing ν and α by P .

Source: Data generated from the EES software developed by S. A. Klein and F. L. Alvarado. Original sources: Keenan, Chao, Keyes, Gas Tables, Wiley, 1984; and Thermophysical Properties of Matter, Vol. 3: Thermal Conductivity, Y. S. Touloukian, P. E. Liley, S. C. Saxena, Vol. 11: Viscosity, Y. S. Touloukian, S. C. Saxena, and P. Hesterman, IFI/Plenum, NY, 1970, ISBN 0-306067020-8.

**FIGURE A-27**

The Moody chart for the friction factor for fully developed flow in circular pipes for use in the head loss relation $\Delta P_L = f \frac{L}{D} \frac{\rho V^2}{2}$. Friction factors in the turbulent flow are evaluated from the Colebrook equation $\frac{1}{\sqrt{f}} = -2 \log_{10} \left(\frac{\Sigma/D}{3.7} + \frac{2.51}{Re \sqrt{f}} \right)$.

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Energiekvationen:

$$\frac{p_1}{\rho g} + \alpha_1 \frac{v_1^2}{2g} + z_1 + h_{\text{pump},u} = \frac{p_2}{\rho g} + \alpha_2 \frac{v_2^2}{2g} + z_2 + h_{\text{turbine},e} + h_L$$

Energiekvationen (KK beteckning):

$$\frac{p_1}{\rho g} + \alpha_1 \frac{v_1^2}{2g} + z_1 + h_p = \frac{p_2}{\rho g} + \alpha_2 \frac{v_2^2}{2g} + z_2 + h_L$$

Bernoullis ekvation: $p_1 + \frac{\rho v_1^2}{2} + \rho g z_1 = p_2 + \frac{\rho v_2^2}{2} + \rho g z_2$

Pumpeffekt: $\eta_{\text{pump}} \dot{W}_{\text{pump}} = \dot{m}gh_{\text{pump}}$

Pumpeffekt (KK): $\eta_p \dot{W}_p = \dot{m}gh_p = \rho \dot{V}gh_p \equiv \dot{V}\Delta p_p = v_{\text{avg}} A_c \Delta p_p$

Kontinuitet volymsflöde: $\dot{V} = v_1 A_1 = v_2 A_2$

Hydrauliska diametern: $D_h = \frac{4A_c}{p}$ $D_h = \frac{2ab}{a+b}$ rektangulärt

Reynolds tal: $\text{Re} = \frac{\rho v D}{\mu} = \frac{v D}{\nu}$ $\begin{cases} \text{Re} < 2300 & \text{laminärt} \\ \text{Re} > 4000 & \text{turbulent} \end{cases}$

Total höjdförlust i rör: $h_L = \left(f \frac{L}{D} + \sum K_L \right) \frac{v^2}{2g}$

Flöde horisontellt cirkulärt rör: $\dot{V} = \frac{(p_1 - p_2)\pi D^4}{128\mu L}$

Tryckförlust i rör: $\Delta p_L = \rho g h_L$

Frikionskoefficient f vid laminär strömning (cirkulärt rör):

$$f = 64/\text{Re}$$

Frikionskoefficient f vid turbulent strömning:

i) Moody: A-27

ii) Colebrook: $\frac{1}{\sqrt{f}} = -2 \log \left(\frac{\epsilon/D}{3.7} + \frac{2.51}{\text{Re}\sqrt{f}} \right)$

iii) Haaland: $\frac{1}{\sqrt{f}} \approx -1.8 \log \left[\frac{6.9}{\text{Re}} + \left(\frac{\epsilon/D}{3.7} \right)^{1.11} \right]$